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NASA TM X-58149
January 1975



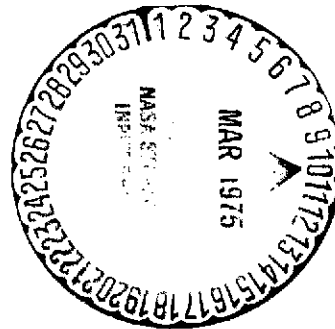
WATER-COOLED FURNACE HEADS FOR USE WITH
STANDARD MUFFLE-TUBE FURNACES

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HEADS FOR USE WITH STANDARD MUFFLE TUBE
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A. J. B.

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
LYNDON B. JOHNSON SPACE CENTER
HOUSTON, TEXAS 77058

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16. Abstract The design of water-cooled furnace seals for use in high-temperature controlled-atmosphere gas and vacuum studies is presented in detailed engineering drawings. Limiting design factors and advantages are discussed.			
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**WATER-COOLED FURNACE HEADS FOR USE WITH
STANDARD MUFFLE-TUBE FURNACES**

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Houston, Texas 77058**

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WATER-COOLED FURNACE HEADS FOR USE WITH
STANDARD MUFFLE-TUBE FURNACES

By Richard J. Williams and O. Mullins*
Lyndon B. Johnson Space Center

SUMMARY

A water-cooled furnace-head system has been designed at the NASA Lyndon B. Johnson Space Center to facilitate studies involving the use of high temperatures in controlled atmospheres. The system incorporates O-ring seals and is useful in noncritical vacuum and gas atmospheres. Although expensive to construct, the design permits adaptation to a wide variety of samples and sensors and has proved to be free of problems during a 2-year period of constant use.

INTRODUCTION

In metallurgical, chemical, and geological studies, sealing ceramic or glass muffle tubes is often necessary so that controlled gas atmospheres or vacuums can be produced and maintained over extended periods. Because such experiments often require the maintenance of high process temperatures, the sealing device must incorporate water cooling. Also, the sealing method should be adaptable to a wide variety of samples and sensors required by various experiments.

FURNACE-HEAD DESIGN

The detailed engineering specifications necessary to build a water-cooled sealing system similar to that designed at the NASA Lyndon B. Johnson Space Center (JSC) are shown in the eight attached engineering drawings. To construct the furnace heads, the following design factors should be considered.

To obtain maximum effect from water cooling, all O-ring grooves should be milled so that all metal parts press tightly against each other. This design enables the running of all processes at a temperature of 1300° C while all parts of the heads are maintained at only slightly warm temperatures.

*Lockheed Electronics Company, Inc.

The end plates were modified for use with a solid ceramic oxygen electrolyte cell and a vertical quenching capacity. End plates can be designed to meet individual experiment requirements.

Despite their heavy weight, brass heads were used because all joints can be sealed with soft solder. However, aluminum can be used if lighter heads are desired. (Aluminum joints must be heliarc welded.)

All heads should be plumbed for series water flow with the flow running from top to bottom. This method of circulation ensures preheating of the cooling water to prevent thermal shocking of the muffle tube. Any clogging within the series flow system will be noticed immediately because the water outflow will decrease.

The eight drawings are scaled for a muffle tube having an outside diameter of 1.5 inches. This system is the smallest that can conveniently be used with both the ceramic electrolyte cell and vertical quenching. However, the design can be scaled to include larger or smaller muffle tubes if necessary.

The drawings provided should supply investigators with enough data to build furnace-head systems similar to those used at the JSC. Although the heads are expensive, their adaptability provides the investigator with almost complete freedom for experiment design. Several sets of these highly dependable heads have been in almost continuous use for more than 2 years at the JSC without problems.

CONCLUDING REMARKS

The water-cooled furnace-head system designed at the NASA Lyndon B. Johnson Space Center has been in use for 2 years and has proved to be free of problems. Although expensive to construct, the system is adaptable to a wide variety of samples and sensors and permits the investigator complete freedom in experiment design.

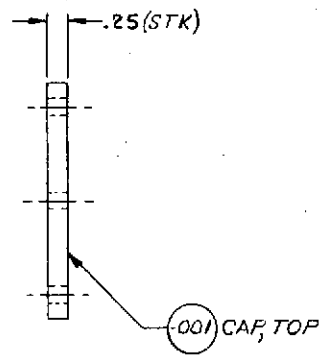
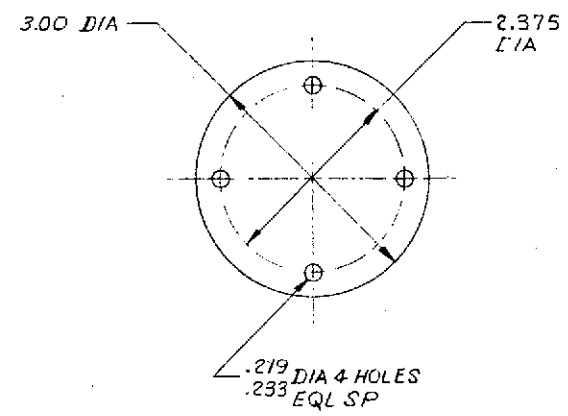
Lyndon B. Johnson Space Center
National Aeronautics and Space Administration
Houston, Texas, January 16, 1975
383-35-00-00-72

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Date 1-21-74 By JLC



1. FABRICATION TOLERANCES AND
PRACTICES PER SKZ36103755.
NOTES: UNLESS OTHERWISE SPECIFIED.

-001		CAP, TOP		BRASS COMP (COML) QQ-B- .25X3.1X3.1 PL 6/3A		APPROV FMSH
PART NUMBER		DESCRIPTION		MATERIAL		SPECIFICATION
DIMENSIONAL TOLERANCE UNLESS NOTED OTHERWISE .0 = .10 .00 = .02 .000 = .005 ANGULAR = —		SIGNATURES DR. R. CLARKSON ENG CH. R. CLARKSON APP. W. P. F. N. AUTH. 27/1/2/4		DATE 9-5-73 9-5-73 11/26/74 12/74		NATIONAL AERONAUTICS & SPACE ADMINISTRATION LYNDON B. JOHNSON SPACE CENTER HOUSTON, TEXAS
SURFACE FINISH IN MICROINCHES RMS UNLESS NOTED OTHERWISE ✓		CAP, TOP— CONTROLLED ATMOSPHERE FURNACE ASSY		CODE IDENT NO. 21356 SIZE C DWS NO. SDB36112459		
SEB36112460		WR4436		SCALE 1/1 ENGRG DIV JN4 SHEET		
NEXT ASSEMBLY						

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- 3 CRAWFORD FITTING COMPANY,
CLEVELAND, OHIO OR EQUAL.
- 2 CAJON COMPANY, CLEVELAND, OHIO
OR EQUAL.
- 1 PARKER SEAL COMPANY, CULVER CITY,
CALIFORNIA OR EQUAL.

NOTES:

1. FABRICATION TOLERANCES AND PRACTICES
PER SKZ 36103755.

NOTES: UNLESS OTHERWISE SPECIFIED.

QTY	PART NUMBER	DESCRIPTION	MATERIAL	SPECIFICATION	ZONE
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8	-501	SCREW CAP, 50C HD	STEEL B-32UNC-2A x 7/16 L	COML	CD PL
2	SDB36112466-001	FLANGE			C3 3
1	SDB36112465-301	CAP ASSY, BOTTOM			B3 4
4	SDB36112464-001	SPACER			D3 3
1	SDB36112463-001	PLUG, QUENCH TUBE			A3 4
1	SDB36112462-001	PROBE TUBE			A4 4
4	SEB36112461-301	BODY ASSY			D3 3
1	SDB36112459-001	CAP, TOP			D3 3
1	-001	FEEDTHRU	MAKE FROM AUT-1-2 FITTING	2	ANW
	-301	CONTROLLED ATMOSPHERE FURNACE ASSY			ANW FUSH

DIMENSIONAL TOLERANCE UNLESS NOTED OTHERWISE		SIGNATURES		DATE	NATIONAL AERONAUTICS & SPACE ADMINISTRATION	
.0	-.10	DIR. CLARKSON		9-4-73	LYNDON B. JOHNSON SPACE CENTER HOUSTON, TEXAS	
.00	-.02	ENG				
.000	-.005	CH E. KANE		9-10-73		
ANGULAR	+ 2°	CR KANE		9-10-73		
SURFACE FINISH IN MICROINCHES RMS UNLESS NOTED OTHERWISE		APP. G. J. J.		11/30/74		
		AUTH. J. J. J.		11/30/74		
					CODE IDENT NO.	SIZE
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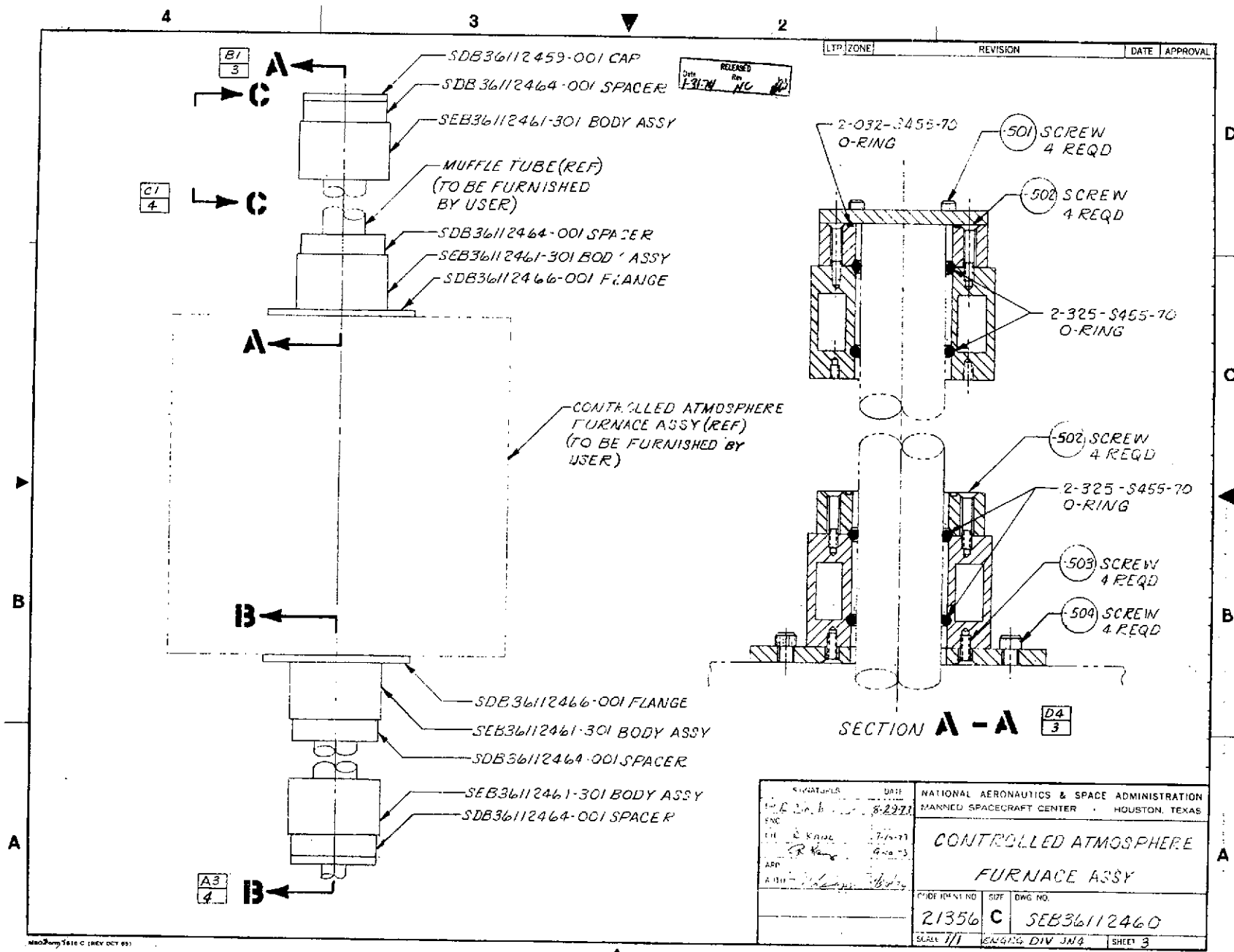
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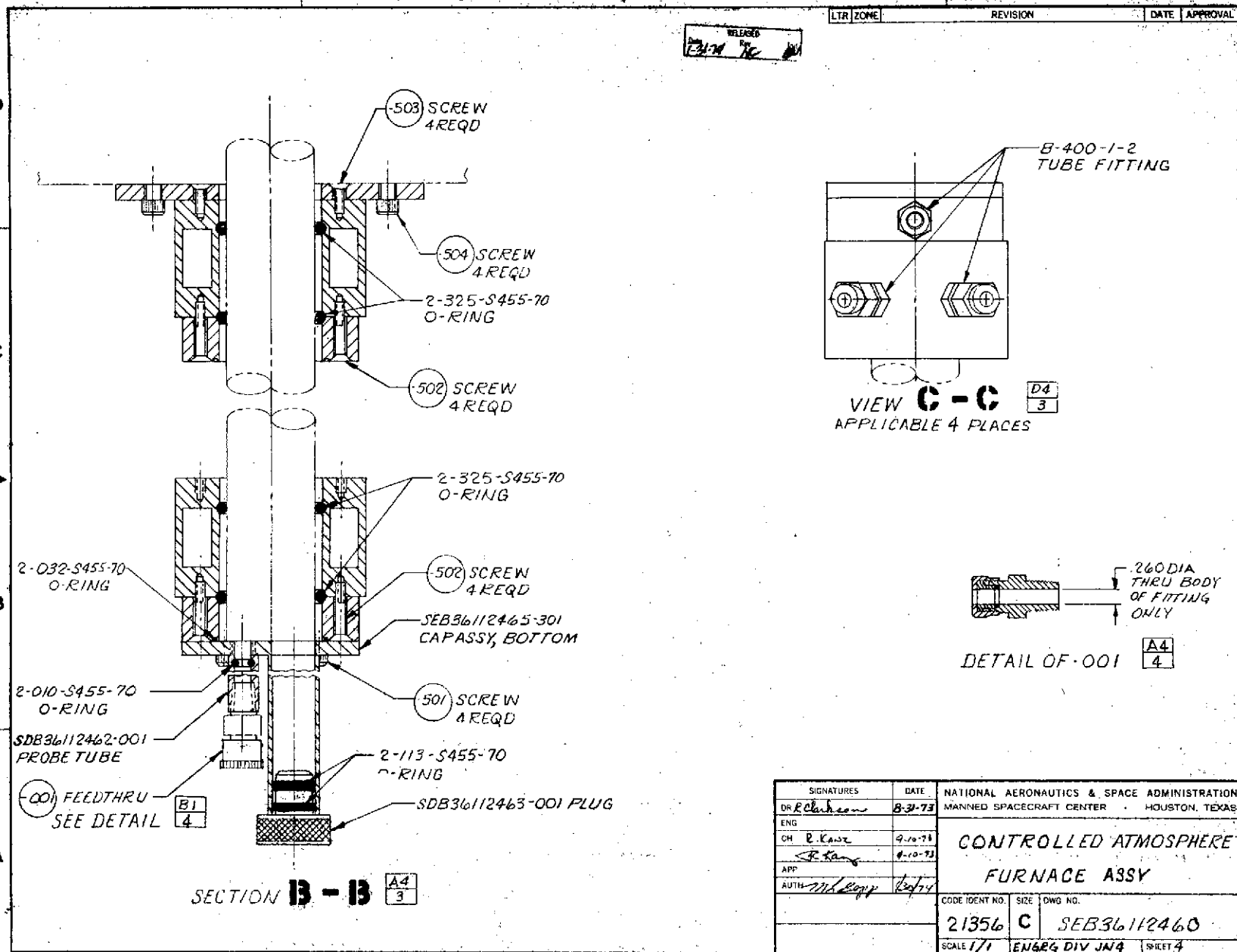
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12	B-400-1-2	TUBE FITTING	3	
8	2-325-S455-70	O-RING	1	
2	2-113-S455-70	O-RING	1	
2	2-032-S455-70	O-RING	1	
1	2-010-S455-70	O-RING	1	
8	-504	SCREW CAP, SOC HD	STEEL 1/4-20UNC-2A x 7/16 L	COML CD FL
8	-503	SCREW CAP, FLH, B2° CSK	STEEL 8-32UNC-2A x 7/16 L	COML BR FASH
QTY	PART NUMBER	DESCRIPTION	MATERIAL	SPECIFICATION
DIMENSIONAL TOLERANCE UNLESS NOTED OTHERWISE .0 ± .10 .000 ± .02 .000 ± .005 ANGULAR ± SURFACE FINISH IN MICROINCHES RMS UNLESS NOTED OTHERWISE		SIGNATURES DR. R. CLARKSON ENG CH R KANE APP AUTH	DATE 9-4-73 9-10-73 9-12-73 7/2/74	NATIONAL AERONAUTICS & SPACE ADMINISTRATION LYNDON B. JOHNSON SPACE CENTER HOUSTON, TEXAS CONTROLLED ATMOSPHERE FURNACE ASSY
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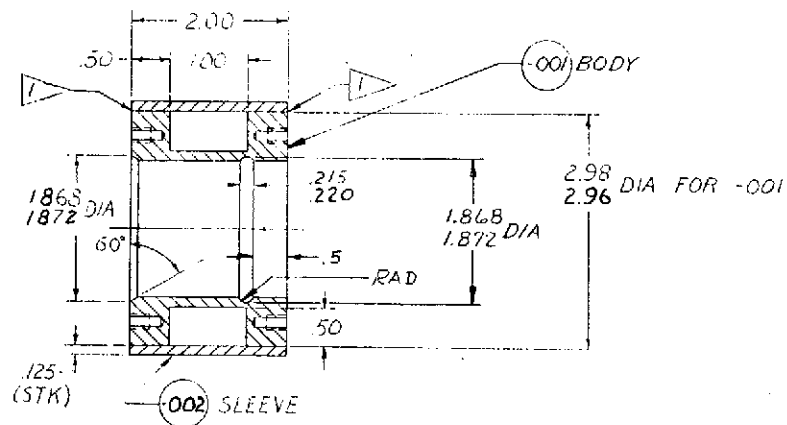
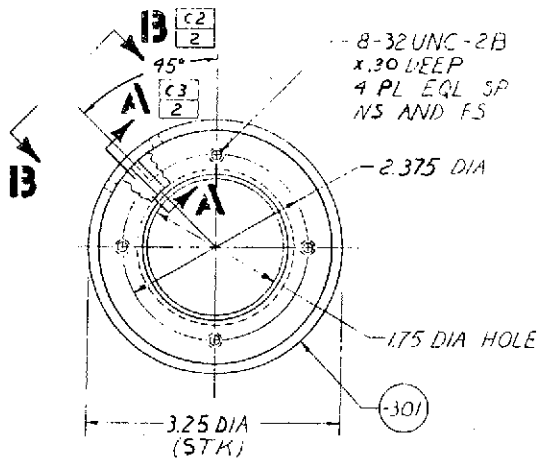


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DR <i>R. Johnson</i>	8-31-73	MANNED SPACECRAFT CENTER	HOUSTON, TEXAS
ENG		CONTROLLED ATMOSPHERE FURNACE ASSY	
CH <i>R. Kase</i>	9-10-73		
APP <i>R. Kase</i>	9-10-73		
AUTH <i>M. B. Bopp</i>	10-1-73		
CODE IDENT NO.	SIZE	DWG NO.	
21356	C	SEB36112460	
SCALE 1/1	ENGRG DIV JN4	SHEET 4	

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2 CUT TO FIT.

1  SILVER SOLDER JOINT USING HIGH TEMPERATURE SOFT SILVER SOLDER PER MANUFACTURERS INSTRUCTIONS.

NOTES.

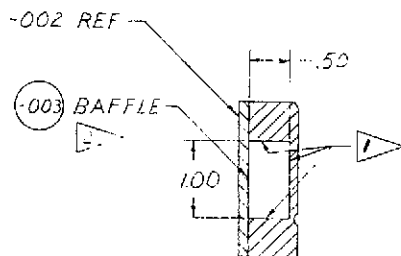
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DIMENSIONAL TOLERANCE UNLESS NOTED OTHERWISE $\phi \quad \pm .10$ $\phi \quad \quad \quad \pm .02$ $.000 \quad \pm .005$ ANGULAR $\pm 2^\circ$	SIGNATURES		DATE		NATIONAL AERONAUTICS & SPACE ADMINISTRATION MANNED SPACECRAFT CENTER • HOUSTON, TEXAS BODY ASSY - CONTROLLED ATMOSPHERE FURNACE ASSY
	DR G. GREEN		8-24-73		
	ENG				
	CH R. CLARKSON		9-6-73		
SURFACE FINISH IN MICRONS RMS UNLESS NOTED OTHERWISE 63 ✓	APR 12 1974		9-10-73		CODE IDENT NO. 21356 SIZE C DWG NO. SEB36112461 SCALE 1/1 ENGRG DIV JN4 SHEET 1 OF 2
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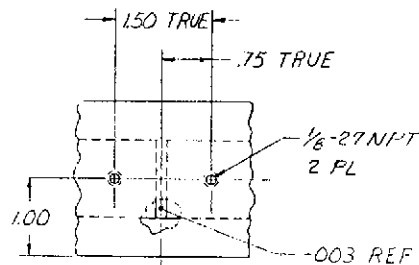
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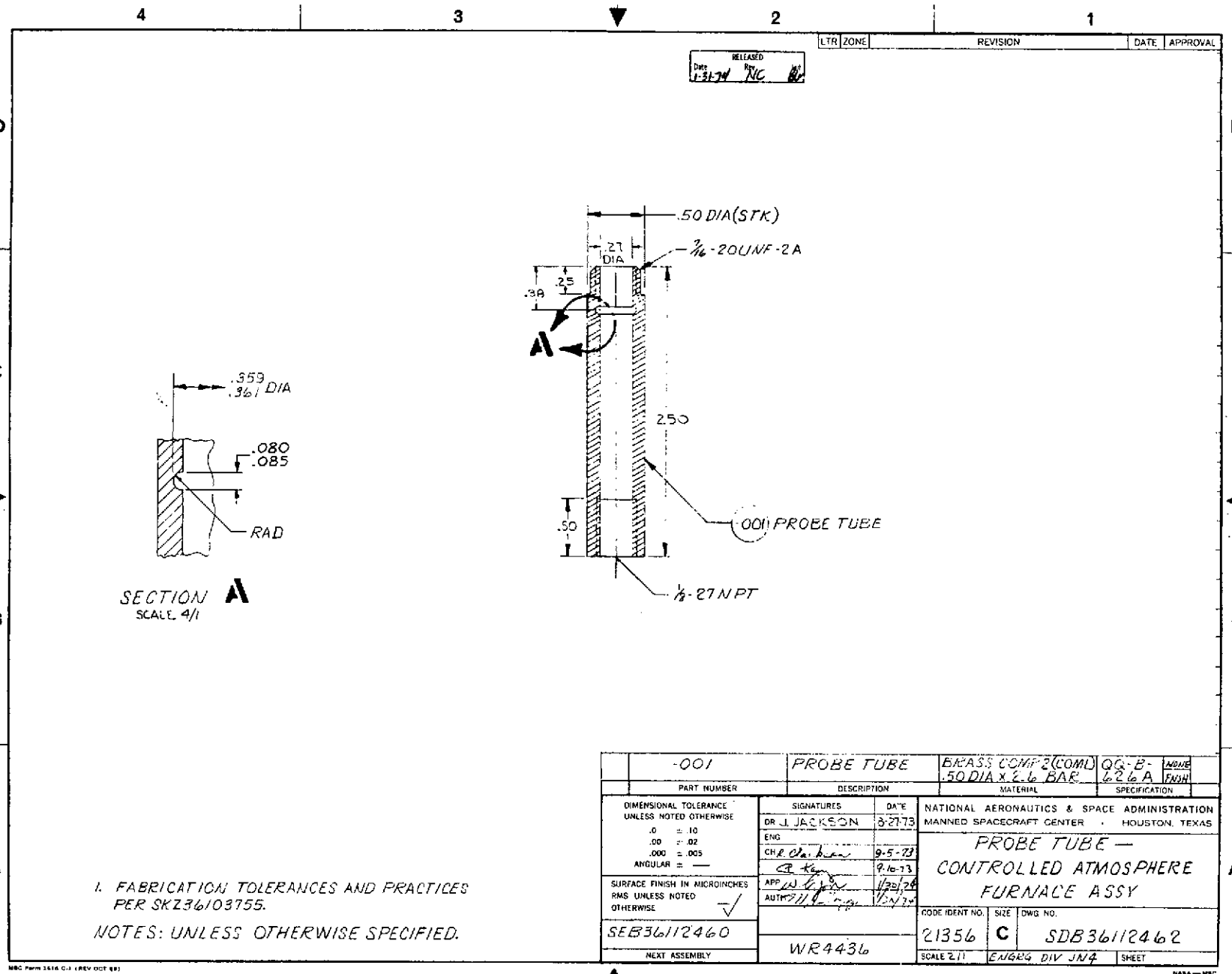
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DR. G. GREEN		8-27-73		
ENG				
CH. L. Markham		9-6-73		
APP				
AUTH 711/3044		1/3/74		
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21356		C	SEB36112461	
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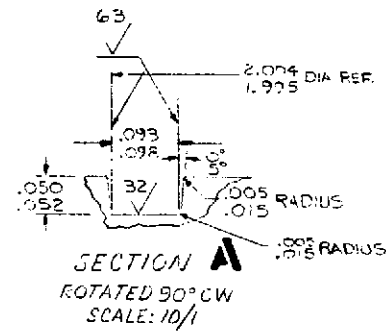
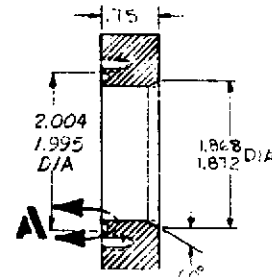
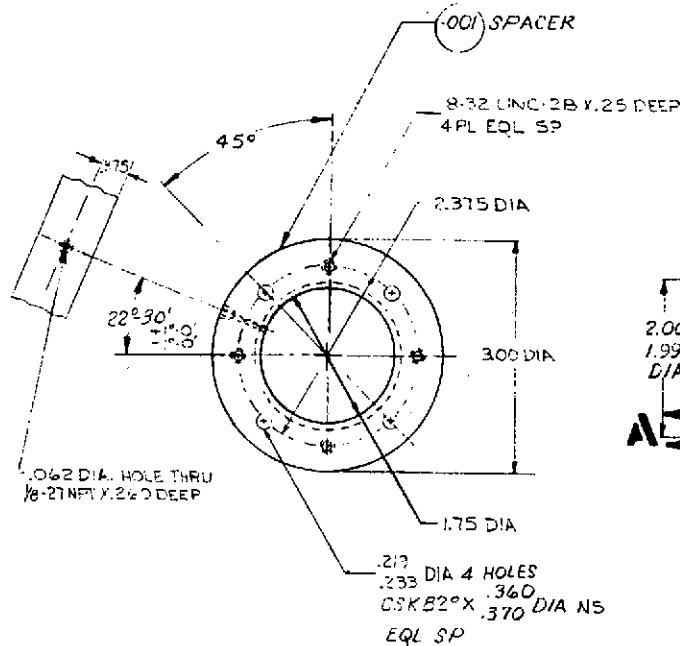
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.00 ± .02		ENG							
.000 ± .005		CH 2 Cl. h. h. h.		9-5-73					
ANGULAR ± —		A. K. K.		9-10-73					
SURFACE FINISH IN MICROINCHES RMS UNLESS NOTED OTHERWISE		APP. W. C. W.		11/20/74					
OTHERWISE		AUTH. 11/20/74		11/20/74					
SEB36112460		WR4436		21356		C		SDB36112462	
NEXT ASSEMBLY				SCALE 2/1		ENGR'G DIV JN4		SHEET	

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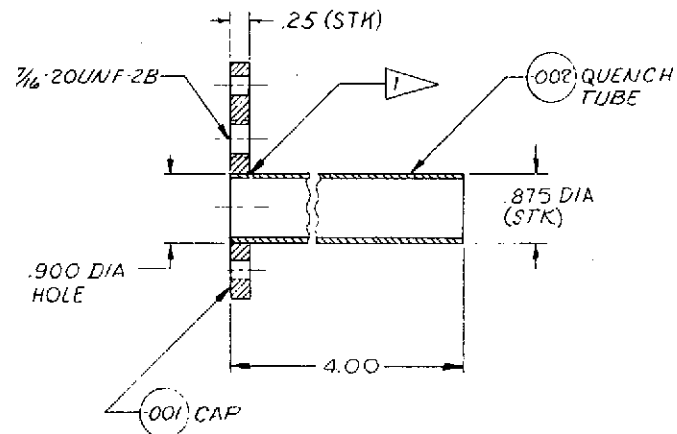
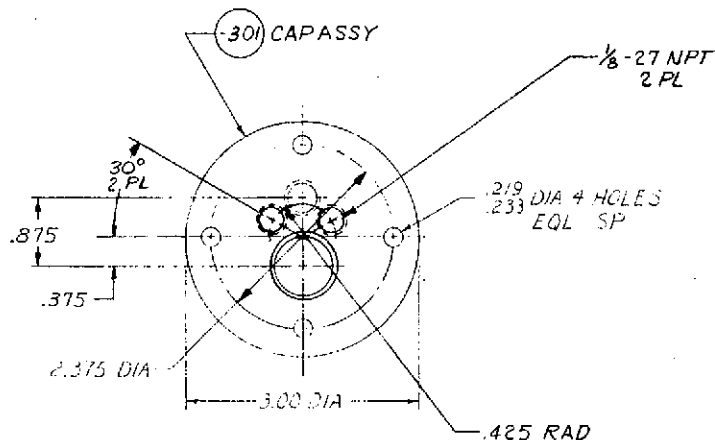
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PER SKZ36103755.
NOTES: UNLESS OTHERWISE SPECIFIED.

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PART NUMBER		DESCRIPTION		MATERIAL		SPECIFICATION	
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SURFACE FINISH IN MICROINCHES RMS UNLESS NOTED OTHERWISE ✓		SEB36112460		SPACER - CONTROLLED ATMOSPHERE FURNACE ASSY			
NEXT ASSEMBLY		WR4436		CODE IDENT NO 21356		SIZE C	
				DWG NO. SDB36112464			
				SCALE 1/1 ENGR G. DIV JNA SHEET			

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1 SILVER SOLDER JOINT USING HIGH TEMPERATURE, SOFT SILVER SOLDER PER MANUFACTURER'S INSTRUCTIONS.

NOTES:

1. FABRICATION TOLERANCES AND PRACTICES PER SK236103155.

NOTES: UNLESS OTHERWISE SPECIFIED.

1	-002	QUENCH TUBE	BRASS ALY NO. 218 7/16 O.D. x .045 WALL x 4.17 LG	COML	NONE
1	-001	CAP	BRASS COMP 2 (COML) .25 x 3.1 x 3.1 PL	QQ-B-613A	NONE
	-301	CAP ASSY, BOTTOM			NONE FINISH

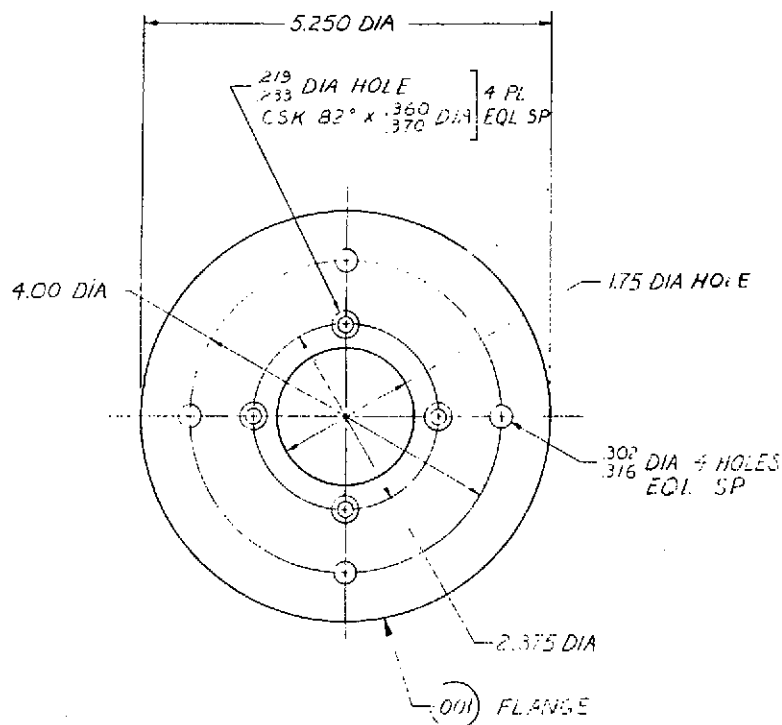
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SURFACE FINISH IN MICROINCHES UNLESS NOTED OTHERWISE ✓	CAP ASSY, BOTTOM - CONTROLLED ATMOSPHERE FURNACE ASSY		
SEB36112460	CODE IDENT NO. 21536 SIZE C DWG NO. SEB36112465		
NEXT ASSEMBLY	SCALE 1/1 ENG'G DIV JN4 SHEET		

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REVISION
21356-46

LTR ZONE REVISION DATE APPROVAL



1. FABRICATION TOLERANCES AND PRACTICES
PER SKZ3-103755.

NOTES: UNLESS OTHERWISE SPECIFIED.

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PART NUMBER	DESCRIPTION	MATERIAL	SPECIFICATION	
	SIGNATURES	DATE	NATIONAL AERONAUTICS & SPACE ADMINISTRATION MANNED SPACECRAFT CENTER • HOUSTON, TEXAS	
	DR G. GREEN	8-23-73	<p>FLANGE - CONTROLLED ATMOSPHERE FURNACE ASSY</p> <p>CODE IDENT NO. 21356 SIZE C DWG NO. SDB36112466</p> <p>SCALE 1/1 ENG'G LVL JN4 SHEET</p>	
	CH R. CLARKSON	9-5-73		
	APP [Signature]	9-10-73		
	AUTH [Signature]	11-22-74		
WR4436				